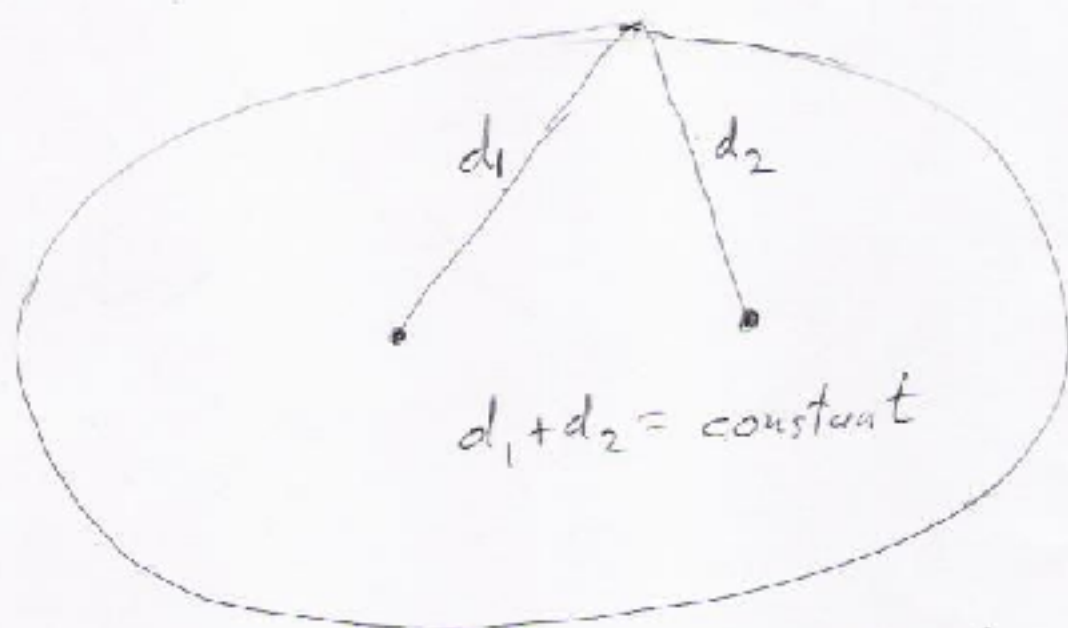


## Ch. 6 Gravity

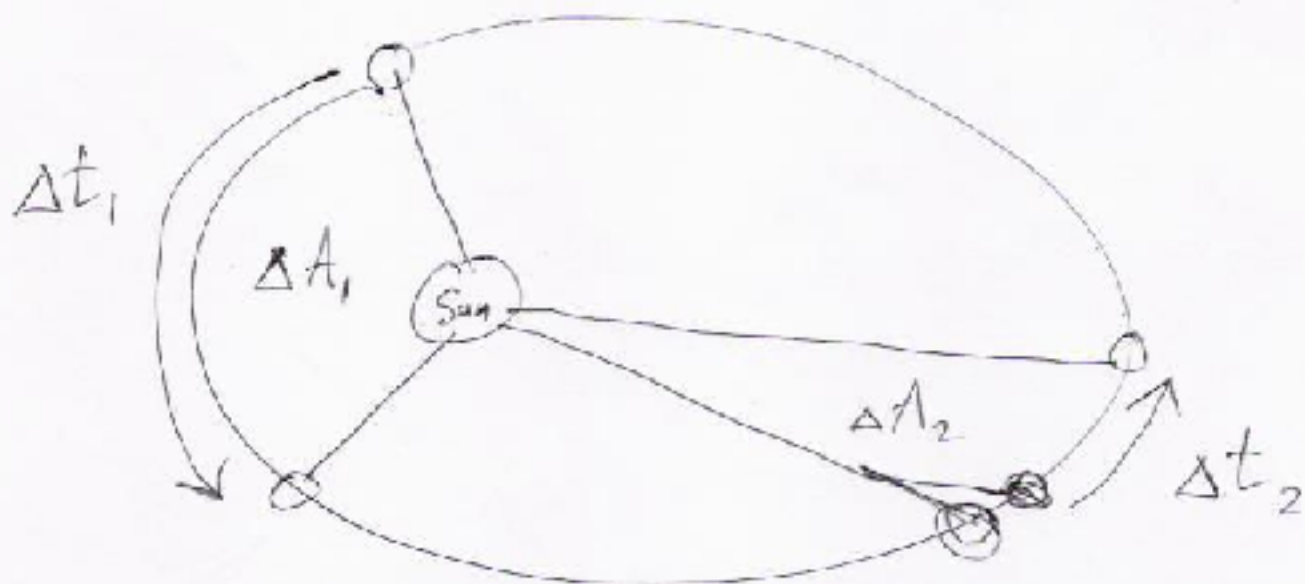
### Kepler's Laws

- 1) Planets orbit the sun in ellipses with the sun at one focus.



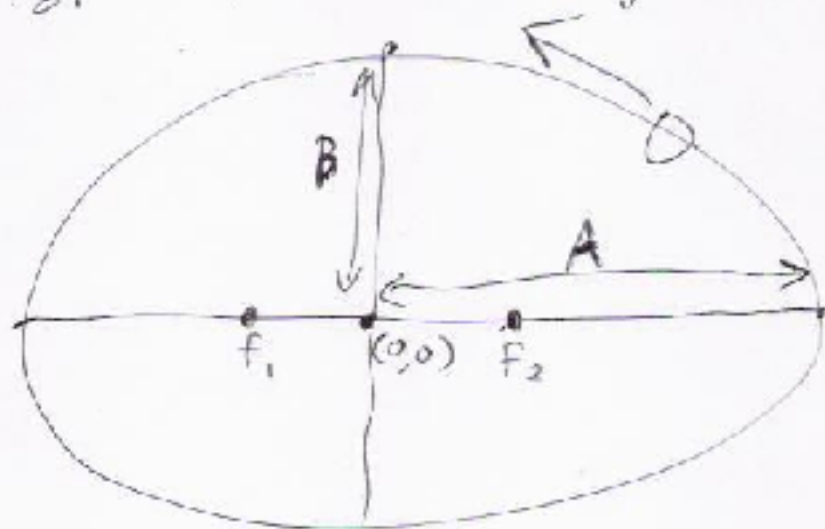
An ellipse is a closed curve s.t. pts on the curve have a constant sum of distances to two points called foci.

2) Planet "sweep out" sectors at a constant rate of area per time.



$$\frac{\Delta A_1}{\Delta t_1} = \frac{\Delta A_2}{\Delta t_2}$$

★ 3) The period of a planet's orbit is proportional to the  $3/2$  power of its semimajor axis.



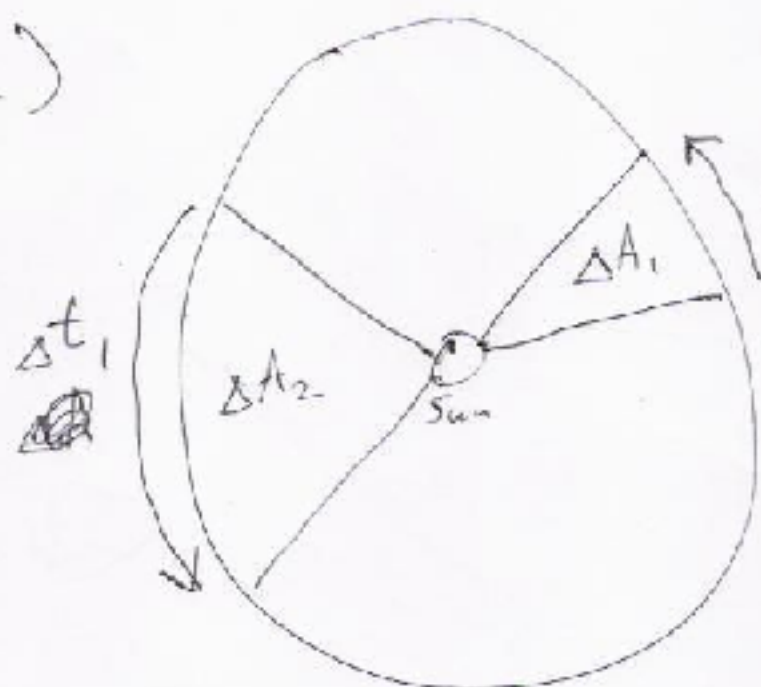
$$T \propto A^{3/2}$$

$\downarrow$   
 is proportional to

# Simplified Kepler's Laws:

1) The planets orbit in circles

2)



$\theta$  in radians  
↓

$$\Delta A_1 = \frac{1}{2} R^2 \Delta\theta_1$$

$$\Delta A_2 = \frac{1}{2} R^2 \Delta\theta_2$$

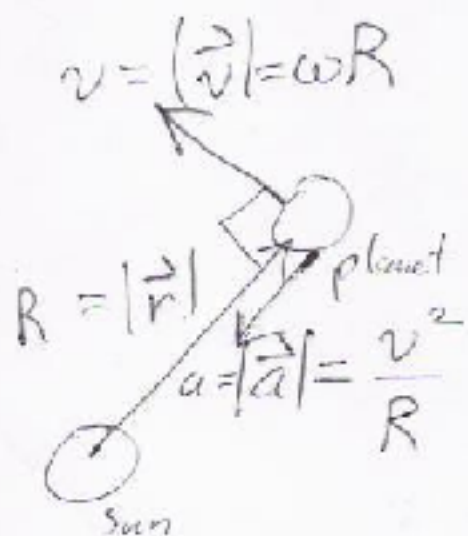
$$\frac{\Delta A_1}{\Delta t_1} = \frac{\Delta A_2}{\Delta t_2}$$

$$\frac{1}{2} R^2 \frac{\Delta\theta_1}{\Delta t_1} = \frac{1}{2} R^2 \frac{\Delta\theta_2}{\Delta t_2}$$

$$\omega = \frac{\Delta\theta_1}{\Delta t_1} = \frac{\Delta\theta_2}{\Delta t_2}$$

$\omega$  is constant.

⇓  
Uniform circular motion.

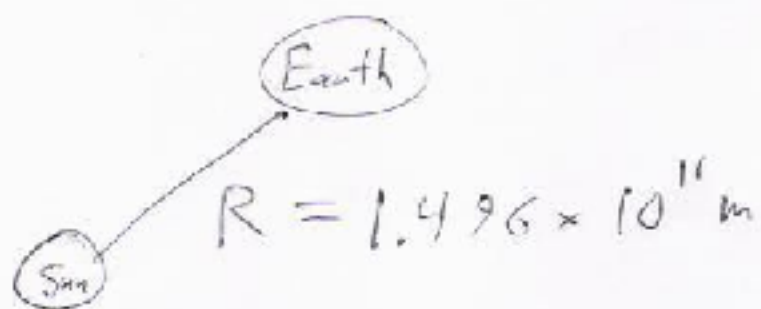


$$v = \omega R$$

$$a = \frac{v^2}{R} = \omega^2 R$$

$$\omega = \frac{2\pi}{T}$$

$T = \text{period}$



$$T = 365.25 \text{ days} = 3.156 \times 10^7 \text{ s}$$

$$\omega = \frac{2\pi}{T} = 1.99 \times 10^{-7} \text{ rad/s}$$

$$\omega = \frac{360^\circ}{T} = 1.141 \times 10^{-5} \text{ day/s}$$

$$v = \omega R = 2.978 \times 10^4 \text{ m/s}$$

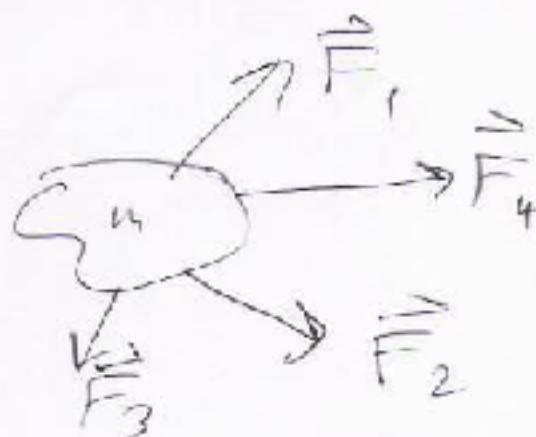
$\frac{2\pi}{360} = \frac{\pi}{180}$

$$a = \frac{v^2}{R} = \omega^2 R = 5.929 \times 10^{-3} \text{ m/s}^2$$

Newton

$$\Sigma \vec{F} = m \vec{a}$$

↑  
inertial mass



## Electrical force



$$F_E = \frac{k|q_1 q_2|}{R^2}$$

## Gravitational force



$$F_G = \frac{G m_1 m_2}{R^2}$$

$k, G$  constants

$q_1, q_2$  electrical charges

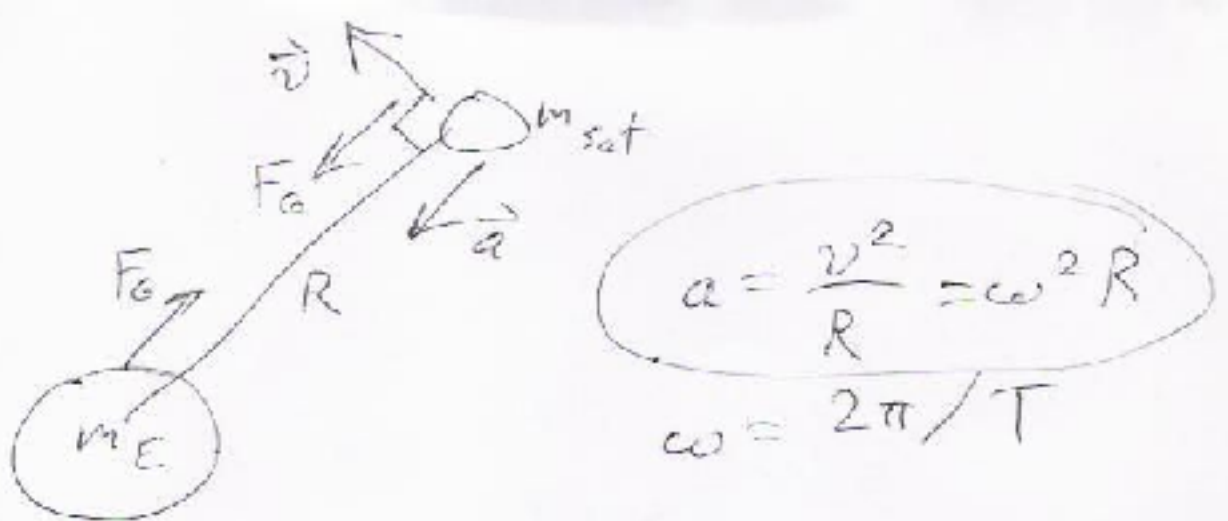
$m_1, m_2$  gravitational charges

inertial mass = gravitational charge  
↑

Gravity & curved spacetime coordinate systems are the same thing. - Einstein

(Galaxies curve path of light.)

At what radius from the earth's center would a satellite orbit in a circle with a period of 12 hours? What would its speed & acceleration be?



$$a = \frac{v^2}{R} = \omega^2 R$$

$$\omega = 2\pi / T$$

$$m_{sat} a = F_G = \frac{G m_E m_{sat}}{R^2}$$

$$a = \frac{G m_E}{R^2}$$

$$G = 6.67 \times 10^{-11} \text{ N} \cdot \text{m}^2 / \text{kg}^2$$

$$m_E = 5.97 \times 10^{24} \text{ kg}$$

$$\omega^2 R = \frac{G m_E}{R^2}$$

$$T = 12 \text{ hours} = 43200 \text{ s}$$

↓

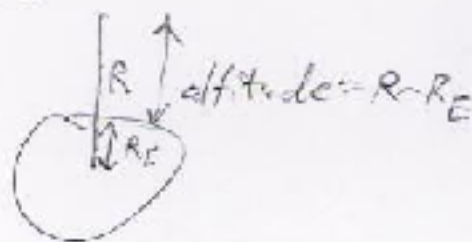
$$\omega = 1.4544 \times 10^{-4} / \text{s}$$

$$\omega^2 R^3 = G m_E$$

$$R^3 = \frac{G m_E}{\omega^2}$$

$$R = \sqrt[3]{\frac{G m_E}{\omega^2}} = 2.66 \times 10^7 \sqrt[3]{\frac{\text{N} \cdot (\text{m}^2 / \text{kg}^2) \cdot \text{kg}}{(\text{1/s})^2}}$$

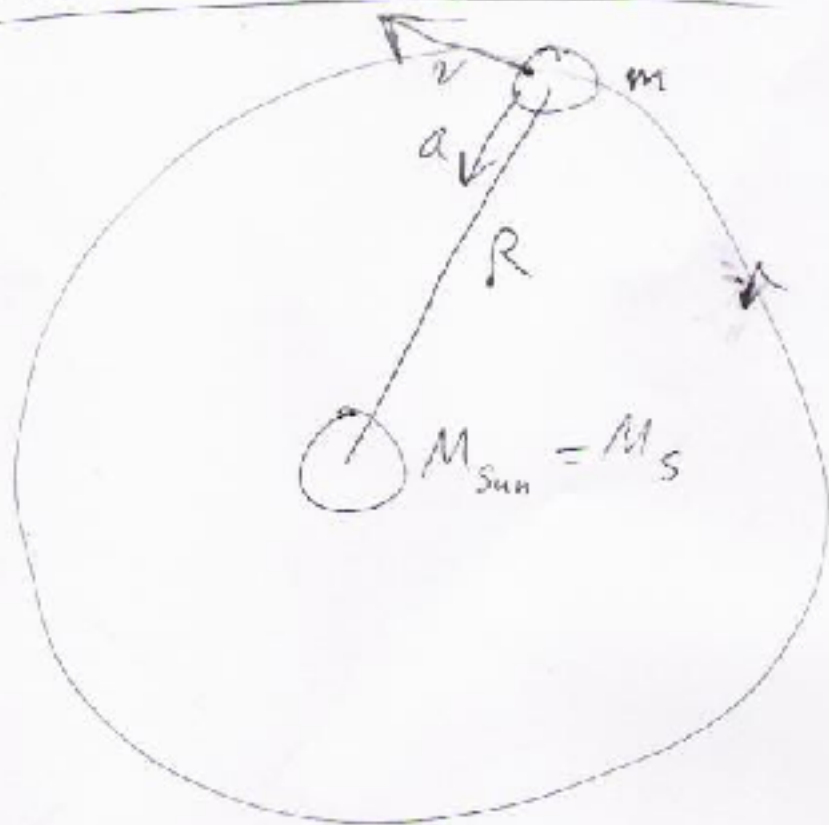
$$R = 2.66 \times 10^7 \text{ m}$$



$$\sqrt[3]{\frac{(\text{kg} \cdot \text{m} / \text{s}^2) (\text{m}^2 / \text{kg}^2) \cdot \text{kg}}{(\text{1/s})^2}}$$

$$\sqrt[3]{\text{m}^3} = \text{m}$$

# Kepler's 3rd Law (For circles)



$$a = \frac{v^2}{R} = \omega^2 R$$

$$\frac{GM_s m}{R^2} = F_G = ma$$

$$\frac{GM_s}{R^2} = a = \omega^2 R$$

$$\omega = \frac{2\pi}{T}$$

$$GM_s = \omega^2 R^3$$

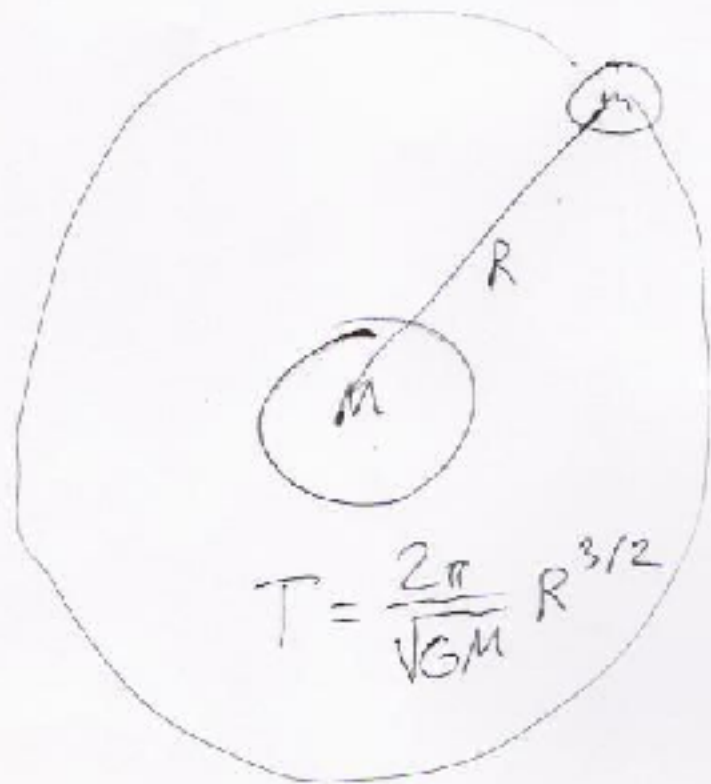
$$GM_s = \frac{4\pi^2}{T^2} R^3$$

$$T^2 GM_s = 4\pi^2 R^3$$

$$T^2 = \frac{4\pi^2 R^3}{GM_s}$$

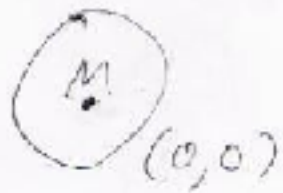
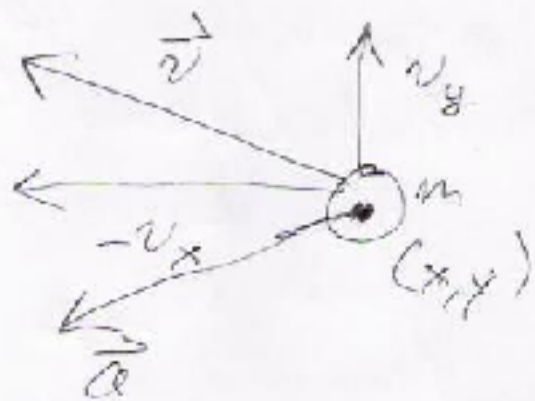
$$T = \frac{2\pi}{\sqrt{GM_s}} R^{3/2}$$

$$T \propto R^{3/2}$$



$$T = \frac{2\pi}{\sqrt{GM}} R^{3/2}$$

# Simulating elliptical orbits:



$$ma = F_G = \frac{GMm}{R^2}$$

$$R^2 = x^2 + y^2$$

$$\hat{r} = \frac{\vec{r}}{|\vec{r}|} = \frac{(x, y)}{\sqrt{x^2 + y^2}}$$

$$a = \frac{GM}{x^2 + y^2}$$

$$\vec{a} = -\hat{r} = \frac{-(x, y)}{\sqrt{x^2 + y^2}}$$

$$\vec{a} = |\vec{a}| \hat{a} = \frac{GM}{x^2 + y^2} \cdot \frac{-(x, y)}{\sqrt{x^2 + y^2}}$$

$$a_x = \frac{-GMx}{(x^2 + y^2)^{3/2}} \quad a_y = \frac{-GM y}{(x^2 + y^2)^{3/2}}$$

$$dv_x = a_x \cdot dt \quad dx = v_x \cdot dt$$

$$v_x = v_x + dv_x \quad x = x + dx \quad t = t + dt$$