

$$a_A = a_B$$

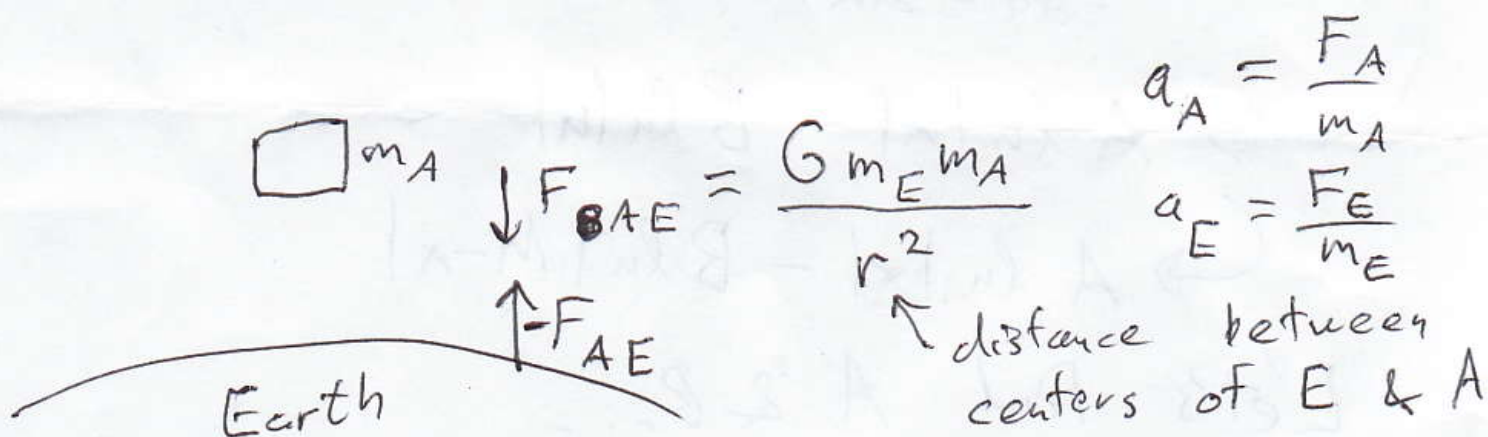
$$F = \frac{dp}{dt} = \frac{d(mv)}{dt} = m \frac{dv}{dt} = ma$$

if  $m$  constant

~~downward~~  $F_A = \frac{dp_A}{dt} = m_A a_A$  (downward)

$$F_B = \frac{dp_B}{dt} = m_B a_B$$
 (upward)

Forces come in pairs. They are interactions between pairs of objects.



$$a_A = a_B = a$$

$$m_A a_A = F_A \text{ (downward)} = m_A g - F_T$$

$$m_B a_B = F_B \text{ (upward)} = F_T - m_B g$$

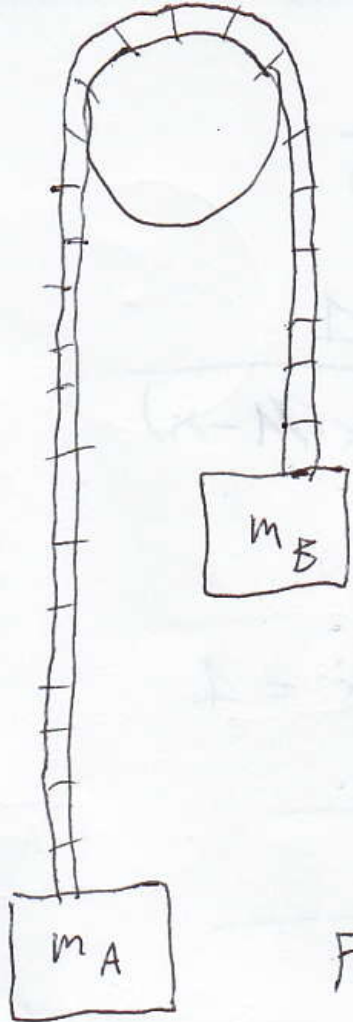
$$\begin{cases} m_A a = m_A g - F_T \\ m_B a = F_T - m_B g \end{cases}$$

Solve for  $a, F_T$ .

$$m_A a + m_B a = m_A g - F_T + F_T - m_B g$$

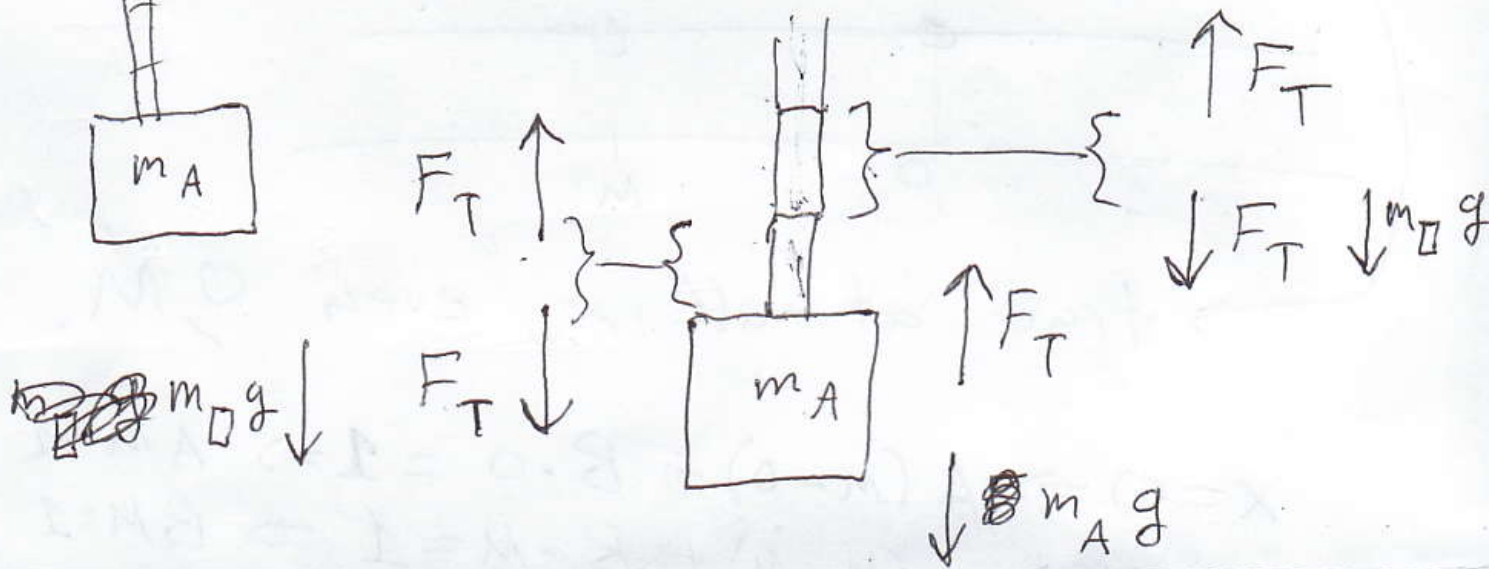
$$(m_A + m_B) a = (m_A - m_B) g$$

$$a = \frac{m_A - m_B}{m_A + m_B} g$$

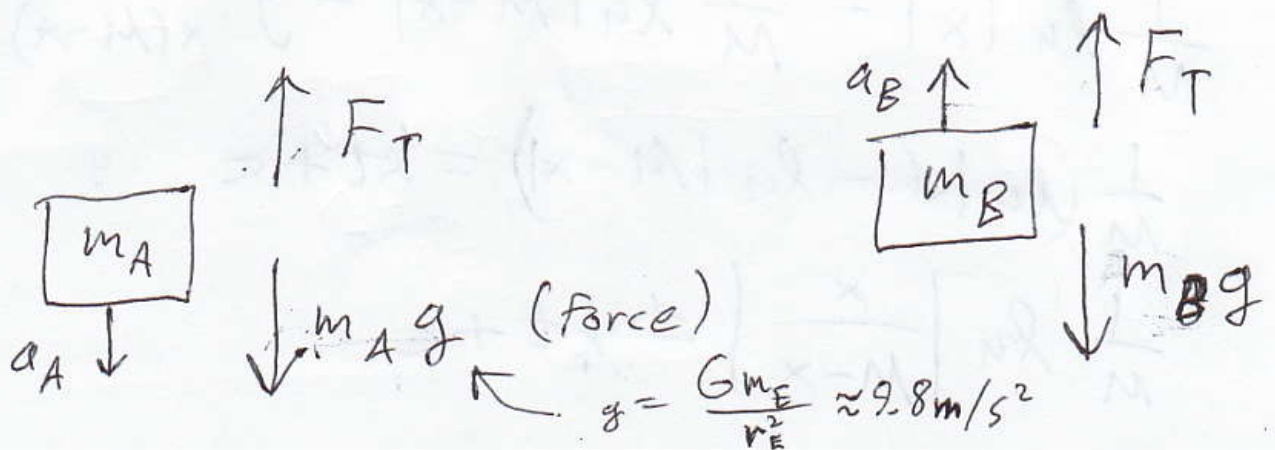


Each piece exerts a ~~to~~ force on each of its neighbors, so that they stay connected.

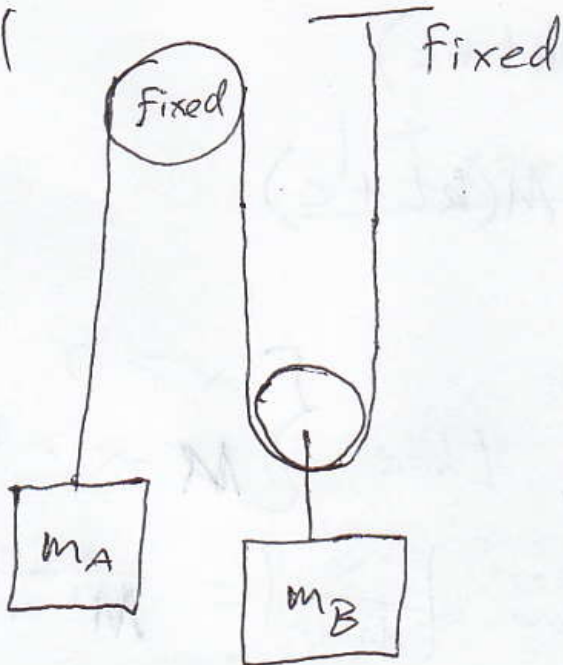
Call this force  $F_T$   
(T for tension)



Ignore mass of rope



HW#1



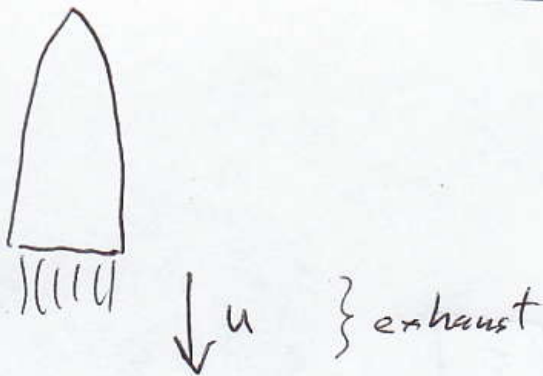
$$m_A = 5.0 \text{ kg}$$

$$m_B = 3.0 \text{ kg}$$

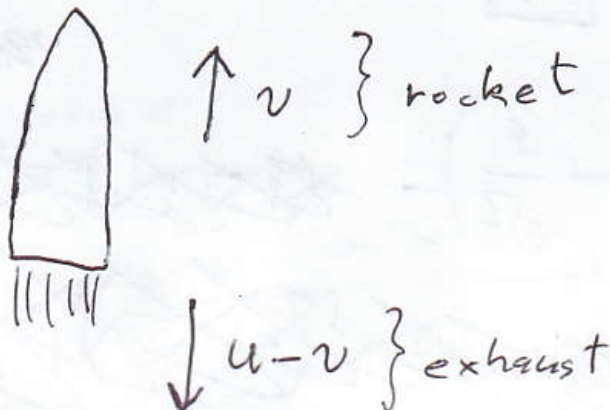
Find the acceleration of  $m_A$  and of  $m_B$ .

What directions are they in?

rocket frame:



Earth frame:



Assume initial rocket mass  $m_0 = 1.2 \times 10^5 \text{ kg}$ .

$u =$  exhaust velocity downward relative to rocket

Assume  $u = \cancel{2.0 \times 10^3} 2.4 \times 10^3 \text{ m/s}$  is constant

$$\frac{dp_{\text{rocket}}}{dt} = \underbrace{-m_{\text{rocket}} g}_{\substack{\text{Force of gravity} \\ \text{on rocket}}} + \underbrace{\left(-\frac{dm_{\text{rocket}}}{dt}\right)}_{\substack{dm_{\text{exhaust}} \\ dt}} \underbrace{(u - v_{\text{rocket}})}_{\substack{v_{\text{exhaust}} \\ \text{downward}}} \text{ frame} \leftarrow \text{Earth's}$$

Upward momentum's rate of change

rate upward momentum transferred to rocket due to gravity

rate  $v$  momentum transferred to downward exhaust gas due to ~~from~~ burning

~~include sign~~  
 upward rate momentum transferred to rocket due to burning

("-" sign because gravity pulls down)

$$g = \frac{G m_E}{r^2}$$

$r =$  distance from rocket to earth's center

For simplicity, ~~we'll~~ we'll just study the rocket's flight when  $r$  is not too different from  $r_E$ , that is, when the rocket's altitude is small compared to  $r_E = 6.4 \times 10^6 \text{ m}$ .

So,  $g \approx \frac{Gm_E}{r_E^2} = 9.8 \text{ m/s}^2$

Assume initial speed  $v_0$  is 0.

Assume  $\frac{dm_{\text{rocket}}}{dt} = -1.0 \times 10^3 \text{ kg/s}$

is constant

Call  $w = 1.0 \times 10^3 \text{ kg/s} = -dm_{\text{rocket}}/dt$

$$\frac{dp_{\text{rocket}}}{dt} = m_{\text{rocket}}g + w(u-v)$$

~~$m_{\text{rocket}} = \int_{t=0}^{t=t} w dt = -w(t-0)$~~

Because  $w$  is constant,

$$\frac{m - m_0}{t - 0} = \frac{\Delta m}{\Delta t} = -w; \text{ so } m - m_0 = -w(t - 0),$$

so  $m = m_0 - wt$ .

(To save space, I've stopped using "rocket.")

~~$$m \frac{dv}{dt} = \frac{dp}{dt} + wv = mg + wv$$~~

~~$$(m_0 - wt) \frac{dv}{dt}$$~~

$$\frac{dp}{dt} = -mg + \cancel{wv} = \boxed{-(m_0 - wt)g + w(u - v)}$$

$$\frac{dp}{dt} = \frac{d(mv)}{dt} = \frac{dm \cdot v + m \cdot dv}{dt}$$

$$\Rightarrow \frac{dp}{dt} = \frac{dm}{dt} v + m \frac{dv}{dt} = -wv + m \frac{dv}{dt}$$

$$\Rightarrow \frac{dp}{dt} = \boxed{-wv + (m_0 - wt) \frac{dv}{dt}}$$

~~$$-wv + (m_0 - wt) \frac{dv}{dt} = (wt - m_0)g + wu - wv$$~~

$$\frac{dv}{dt} = \frac{(wt - m_0)g}{m_0 - wt} + \frac{wu}{m_0 - wt}$$

$$\frac{dv}{dt} = -g + \frac{wu}{m_0 - wt}$$

$$dv = -g dt + \frac{wu dt}{m_0 - wt}$$

$$v = v - 0 = v - v_0 = \int_{v_0}^v dv = \int_{t_0}^t -g dt + \int_{t_0}^t \frac{wu dt}{m_0 - wt}$$

$$v = -g(t - \underbrace{t_0}_{=0}) + \int_{\underbrace{0}_{v_0}}^t \frac{wu dt}{m_0 - wt} = -gt + wu \int_0^t \frac{dt}{m_0 - wt}$$

$$v = -gt + wu \int_{t_0}^t \frac{dt}{m_0 - wt}$$

$$m = m_0 - wt$$

$$dm = 0 - w dt$$

$$-dm/w = dt$$

$$v = -gt + wu \left( -\frac{1}{w} \ln \frac{m}{m_0} \right)$$

$$-\frac{1}{w} \ln \frac{m}{m_0}$$

$$v = -gt - u \ln \frac{m}{m_0}$$

$$v = -gt + u \ln \frac{m_0}{m}$$

In general, if  $0 < a < b$ ,

$$\text{then } \int_a^b \frac{dx}{x} = \ln \frac{b}{a}.$$

$$v = -gt + u \ln \left( \frac{m_0}{m_0 - wt} \right)$$

For example, if  $t = 90s$ , then

$$v = 2.4 \times 10^3 \text{ m/s.}$$

$$\text{If } t = 100s, \quad v = 3.3 \times 10^3 \text{ m/s.}$$

( $0 \leq t < 120s$  because in 120s, all the mass of the rocket would be gone, burned into exhaust gas, so before then we run out of fuel to burn.)

$$120s = \frac{m_0}{w} \text{ is how I found } 120s.)$$